



Rule Based Aircraft Performance Systems

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RULE BASED AIRCRAFT PERFORMANCE SYSTEMS

ABSTRACT

For aircraft such as BOEING and AIRBUS, there are many rules for takeoff and landing performance calculations. These rules are defined according to aircraft configuration, MEL (Minimum Equipment List) / CDL (Configuration Deviation List) items, Turn Procedures (if available), Airport and Obstacle Database (AODB) and Weather Conditions.



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ABSTRACT

In this study, a general database is created to include all kinds of aircraft, airport information and rules effecting aircraft performance. This database system is server based and can be accessed from everywhere. The rules include aircraft configuration such as airframe/engine combination, flap setting, airconditioning, anti-ice, thrust rating, aircraft CG position, climb method etc. , MEL/CDL items and external conditions such as AODB and weather conditions such as runway condition, wind direction, wind speed, outside air temperature, QNH and NOTAMs (if available). Rules can be simple or complex in such a way that composite conditions can be used all together and simple rules can be combined in a complex rule.



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ABSTRACT

This study is different from other studies in that all aircraft types can be combined in a single server based database systems and rule checking process is very dynamic and flexible. These rules can be managed easily on server. This server database can be extended to create national or international aircraft and airport information system in the future.



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ABSTRACT

Besides the advantages of operational safety resulting from making point calculation instead of conventional calculations, with the implementation of this system fuel conservation, emissions of CO₂ decrement, aircraft engine health management are achieved.



PREVIOUS RELATED RESEARCHES

A prototype rule-based front end expert system for integrity enforcement in relational data bases: An application to the naval aircraft flight records data base

*Expert Systems with Applications, Volume 8, Issue 1, January-March 1995,
Pages 47-58
Magdi N. Kamel*

The goal of this study is to accomplish the accuracy of the real world data in the databases in accordance with organisational procedures.



PREVIOUS RELATED RESEARCHES

Lateral Flight Technical Error Estimation Model for Performance Based Navigation

Chinese Journal of Aeronautics, Volume 24, Issue 3, June 2011, Pages 329-336

Hongsheng ZHAO, Xiaohao XU, Jun ZHANG, Yanbo ZHU, Chuansen YANG, Sheng HONG

The goal of this study is to increase environmental turbulence intensity in Performance Based Navigation (PSN) to decrease Flight technical error (FTE) combined with navigation system error (NSE).



PREVIOUS RELATED RESEARCHES

Rule-based optimization approach for airline load planning system

*Procedia Computer Science, Volume 1, Issue 1, May 2010, Pages 1455-1463
Feng Li, Chunhua Tian, Hao Zhang, Wade Kelley*

The goal of this study is to increase ground operations efficiencies of an American airline by making rule based load planning.



PREVIOUS RELATED RESEARCHES

A rule-based reactive model for the simulation of aircraft on airport gates

*Knowledge-Based Systems, Volume 10, Issue 4, January 1998, Pages 225
236*

Yu Cheng

The goal of the study is to resolve the conflict of time and apron traffic to increase the apron control activity.



PREVIOUS RELATED RESEARCHES

Using decision rules to achieve mass customization of airline services

*European Journal of Operational Research, Volume 205, Issue 3, 16
September 2010, Pages 680-686*

James J.H. Liou, Leon Yen, Gwo-Hshiung Tzeng

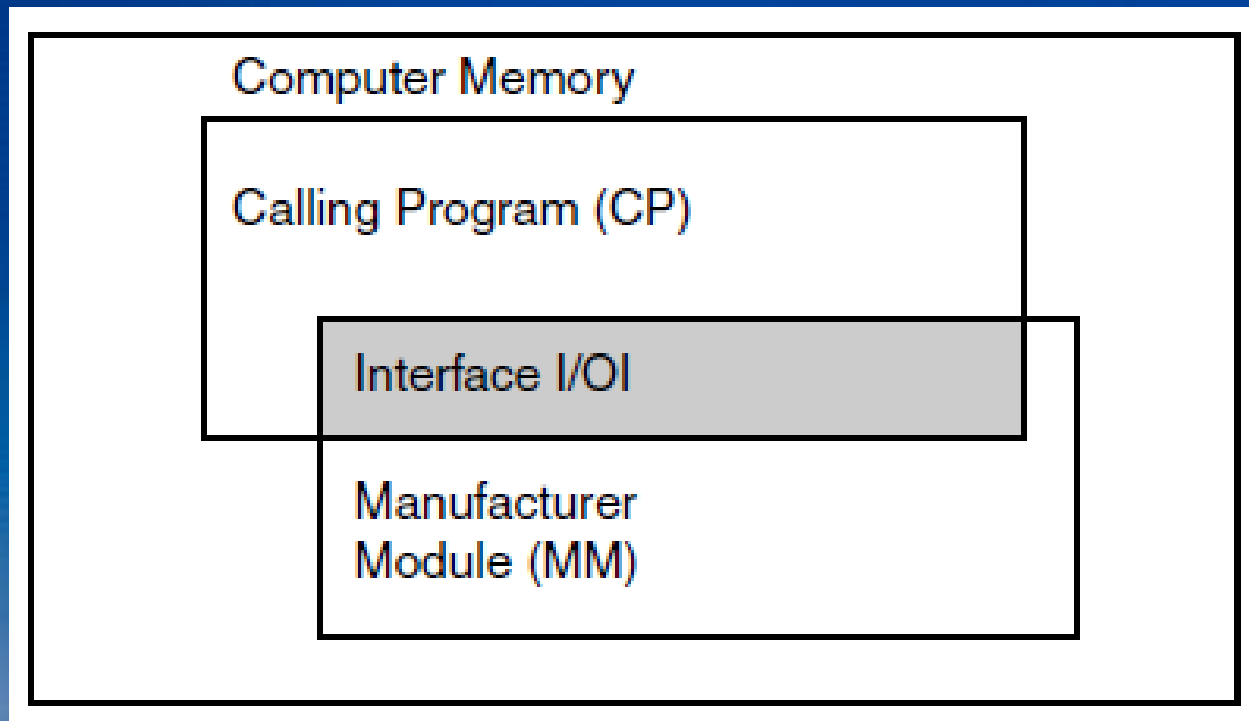
The goal of this study is to increase airlines service quality and gain additional revenue without effecting the passenger perception.



RULE BASED AIRCRAFT PERFORMANCE SYSTEMS

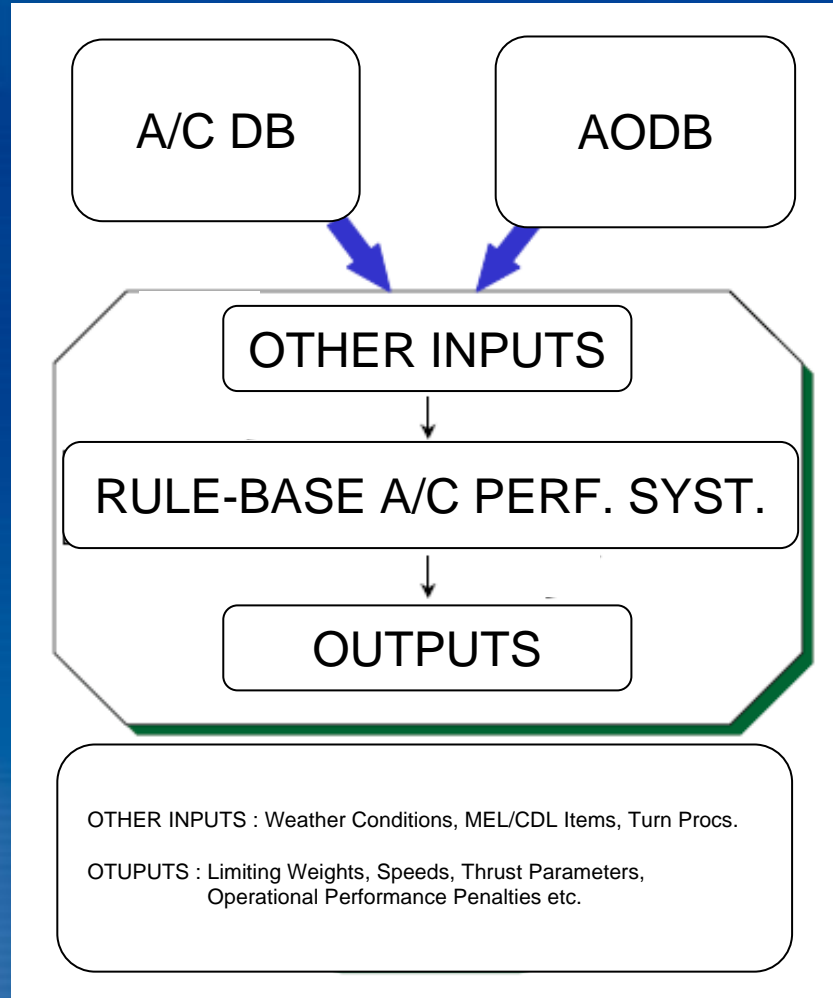
AIRCRAFT PERFORMANCE SYSTEMS

INTERFACE DIAGRAM



RULE BASED AIRCRAFT PERFORMANCE SYSTEMS

AIRCRAFT PERFORMANCE SYSTEMS



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AIRCRAFT PERFORMANCE SYSTEMS

EFB Performance Modules
✖

Takeoff
Landing
Weight and Balance

Aircraft Information

A/C Registration: **TC-SKH**

A/C Type: **B737-800**

Engine Type: **CFM56-7B26**

Aircraft Information

Aircraft Configuration

Flap Setting: **OPTIMUM** MEL / CDL Item: **NO**

Airconditioning: **AUTO** CG Position:

Anti - Ice: **OFF**

Alternate FWD CG: **FORWARD**

Improved Climb: **NONE OR FIXED**

Thrust Rating: **26K (FULL RATED)**

Takeoff Weight

MAXIMUM ACTUAL

67000 Kg

Airport Information

Airport IATA Code: **AYT** Search Runway: **36C** RWY DSG

Airport Name: **ANTALYA** Elevation: **176 Ft**

Line-up Position: **ALL** TORA: **11155 Ft**

TODA: **11352 Ft** ASDA: **11155 Ft**

Slope TORA: **0.00 %** Slope ASDA: **0.00 %**

ATIS Information

Runway Condition: **DRY** NOTAM: **NO**

Wind Direction: **350 °** Wind Speed: **9 Kt**

Headwind: **9 Kt** Crosswind: **2 Kt**

OAT: **17 °C** QNH: **1013.25 HPa**

Weight Limitations

Field Length: **86182 Kg**

Obstacle **OBS**: **85669 Kg**

Climb: **84418 Kg**

Brake Energy: **83757 Kg**

Tire Speed: **86182 Kg**

Certification: **79015 Kg**

RWY Loading: **79015 Kg**

	58 °C	56 °C	54 °C
Assumed TOW	67258 Kg	68344 Kg	69437 Kg
Stop Margin	2147 Ft	2361 Ft	2557 Ft
V1	148 Kts	148 Kts	148 Kts
VR	149 Kts	149 Kts	148 Kts
V2	152 Kts	152 Kts	153 Kts
∑N1	90.18	90.70	91.23

1

FLAP SETTING

976 Ft

M.F.R.A.

79015 Kg

Certification

MATOW

4807 Ft	143 Kts	146 Kts	154 Kts	98.40
STOP MARGIN	V 1	V R	V 2	∑N1
100 Kts				
V MCG	VREF40			143 Kts
5 %	VREF40+30			173 Kts
GRADIENT	VREF40+50			193 Kts
	VREF40+70			213 Kts

Engine Out Procedure

Turn **RIGHT** at D5.0 AYT VOR/DME to 180 degrees. Accelerate and climb to MSA (3000ft)

Special Notes

-Takeoff Thrust Time Limit 10 minutes

View EOP Chart
Analyze
Save
Exit

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CALCULATION PARAMETERS (an example model)

Calculation Input Parameters are pre-defined by IATA for the aircraft manufacturers to establish their values for the related aircraft.

- POPT** - Program Options Array
- CONF** - Aircraft Configuration Array
- XMET** - Meteorological Conditions Array
- RWYD** - Runway Array
- OBSD** - Obstacle Array
- FPTD** - Flight Path Turn Data array
- UNIT** - Units Array
- SPIA** - Supplemental Performance Interface Array



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CALCULATION PARAMETERS (an example model)

XMET - Meteorological Conditions Array (an example array position)

ARRAY POSITION	PARAMETER	VALID VALUES	COMMENT
XMET (1)	TEMPERATURE.		
XMET (2)	WIND COMPONENT		
XMET (3)	CROSSWIND COMPONENT		
XMET (4)	ALTITUDE FLAG	0. = Airport pressurealtitude. 1. = QNH 2. = QFE	
XMET (5)	ALTITUDE VALUE		
XMET (6)	ACTUAL OUTSIDE AIR TEMPERATURE(OAT)		
XMET (7)	TEMPERATURE INVERSION FLAG	0. = No Temperature Inversion 1. = TemperatureInversion	
XMET (8)	TEMPERATURE AT INVERSION ALTITUDE		
XMET (9)	INVERSION ALTITUDE		



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RULES (an example model)

XMET - Meteorological Conditions Array (an example SIMPLE RULE)

ARRAY POSITION	PARAMETER	VALID VALUES	COMMENT
XMET (2)	WIND COMPONENT	VALUE	ALONG THE RUNWAY + HEADWIND - TAILWIND
XMET (5)	AIRPORT ALTITUDE	VALUE	FOR VARIABLE SPECIFIED IN XMET (4)



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RULES (an example model)

POPT - Positions Array (an example COMPLEX RULE)

XMET - Meteorological Conditions Array (an example COMPLEX RULE)

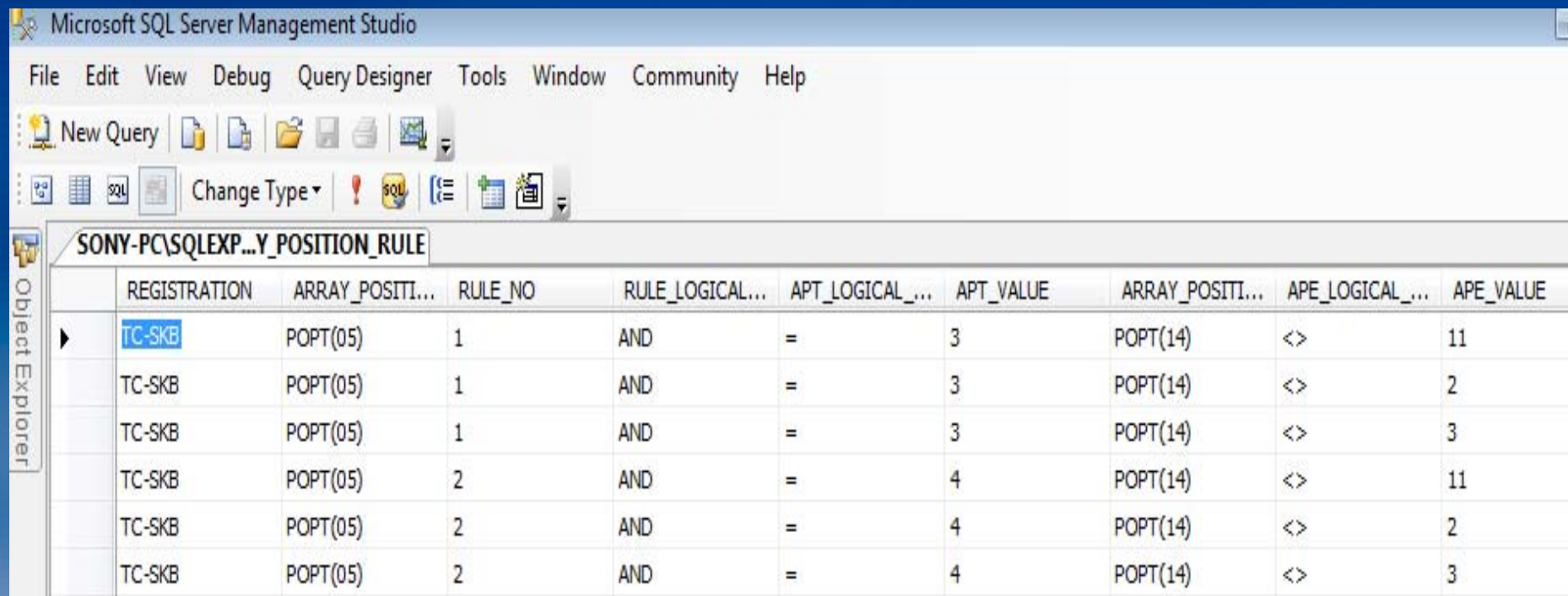
ARRAY POSITION	PARAMETER	VALID VALUES	COMMENT
POPT (14)	SURFACE CONDITION / TYPE OF CONTAMINATION	0 – DRY 1 – WET* 2 – STANDING WATER* 3 – SLUSH* 4 – COMPACTED SNOW * 5 – DRY SNOW* 8 – WET ICE* 11 – ADVISOTY WET	POPT (14) = 2,3,4,5 OR 8 NOT ALLOWED WITH POPT (1) = 4 * WITH DRY CHECK
XMET (6)	ACTUAL OUTSIDE AIR TEMPERATURE (USED FOR MINIMUM CONTROL SPEED AND POWER SETTING CALCULATIIONS)	VALUE OR 9.E20	INPUT IS OPTIONAL FOR POPT (1) SETTINGS OF 0., 1.,OR 2. INPUT IS OPTIONAL FOR POPT (1) SETTINGS OF 4.OR 9.



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DATABASE ARCHITECTURE (an example model)

POPT - Positions Array (an example COMPLEX RULE)



REGISTRATION	ARRAY_POSITI...	RULE_NO	RULE_LOGICAL...	APT_LOGICAL_...	APT_VALUE	ARRAY_POSITI...	APE_LOGICAL_...	APE_VALUE
TC-SKB	POPT(05)	1	AND	=	3	POPT(14)	<>	11
TC-SKB	POPT(05)	1	AND	=	3	POPT(14)	<>	2
TC-SKB	POPT(05)	1	AND	=	3	POPT(14)	<>	3
TC-SKB	POPT(05)	2	AND	=	4	POPT(14)	<>	11
TC-SKB	POPT(05)	2	AND	=	4	POPT(14)	<>	2
TC-SKB	POPT(05)	2	AND	=	4	POPT(14)	<>	3



CONCLUSION

We have concluded that by making a point calculation and taking all of the parameters into account the accuracy of the results have been increased, operational safety is escalated and a user friendly user interface is prepared to decrease both pilot workload in the daily operations and engineering studies workload .

With the server database a national and international aircraft and airport database can be created in the future.

Besides, with the implementation of this system fuel conservation, emissions of CO₂ decrement, aircraft engine health management are achieved.

